

In the Claims:

1. (Original) An aircraft engine, particularly a gas turbine engine, with at least one fan (11) and a core engine (12), whereby the fan (11) comprises a fan housing (13) enclosing a fan flow channel, and at least one fan wheel (15), and whereby the core engine (12) comprises at least one compressor (15, 16), at least one combustion chamber (17), and at least one turbine (18, 19), and with at least one generator (24) for producing electrical energy, whereby the or each generator (24) produces electrical energy by withdrawing shaft power from the core engine (12), characterized in that the or each generator (24), for producing electrical energy, is integrated into at least one strut (21) extending in a radial direction of the fan flow channel, and thus is positioned within the fan flow channel.

2. (Original) The aircraft engine of claim 1, characterized in that the or each generator (24) or the or each strut (21) into which the or each generator (24) is integrated, is demountable out of the fan flow channel for maintenance work.

Claims 3 to 9 (Canceled).